

# SISO Controller of a Bicycle- Rider System

Chad Findlay

Jason Moore

Claudia Pérez-Maldonado

MAE 272 Project, Prof. Joshi

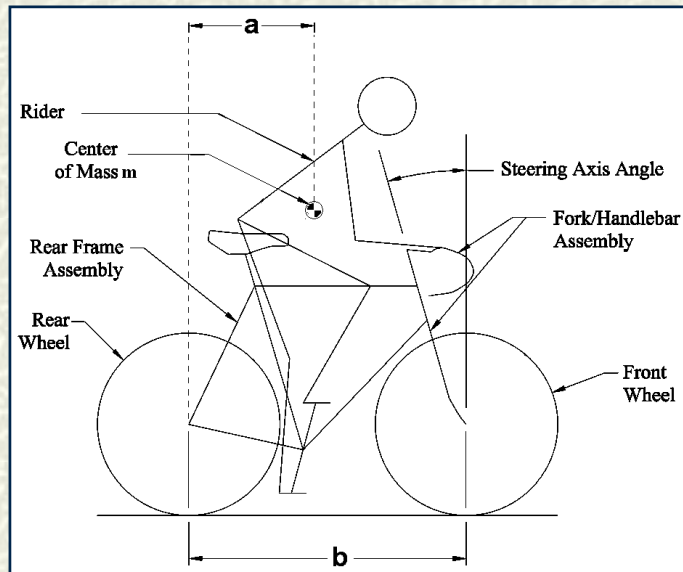
---

# Objectives

---

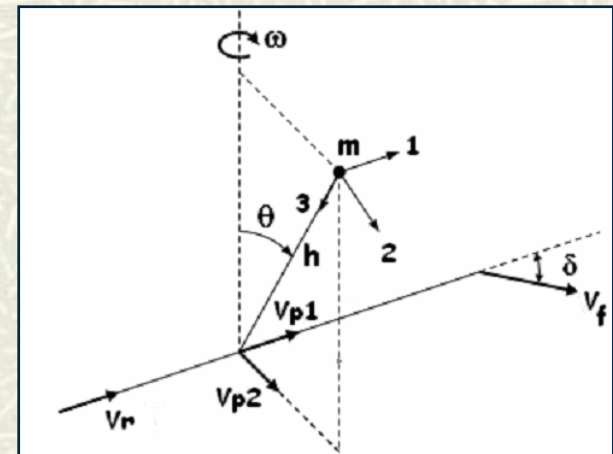
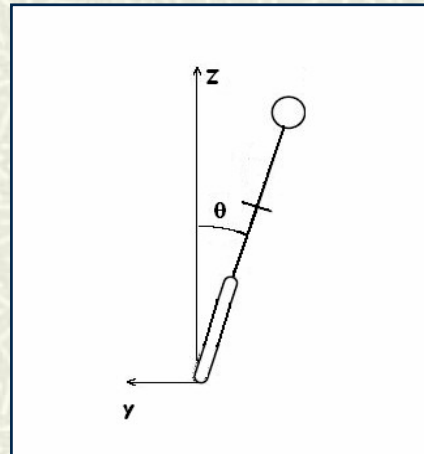
- To design a control system to keep a bicycle-rider system upright by varying only the steering angle.
  - To compare three different control techniques:
    - Pole placement
    - LQR
    - Classical control
-

# Model Description



- Inverted pendulum bicycle model
- One degree of freedom – roll angle ( $\theta$ )
- EOM derived with Lagrange's method
- 2<sup>nd</sup> order system

- $\theta$  = roll angle
- $\delta$  = steering angle



# Assumptions and Linearization

---

- One rigid body representing rider and bicycle
- Negligible mass ideal rolling wheels with no sideslip
- Constant rear wheel velocity
- Inherent stability characteristics due to frame geometry are neglected
- Rolling on a flat level ground plane
- Both the roll and steering angles are assumed to be small

2<sup>nd</sup> Order Equation of Motion:

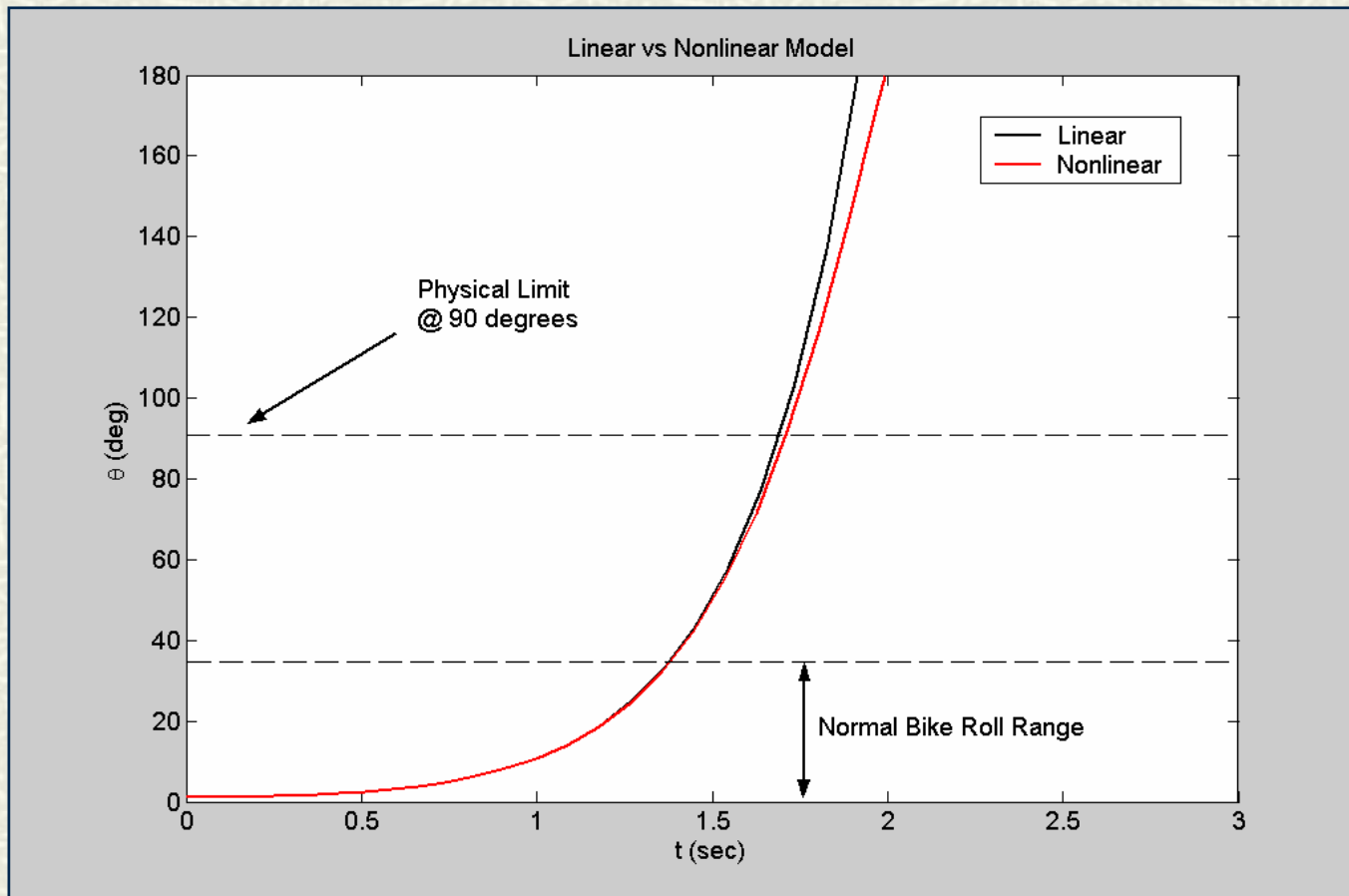
$$(I_1 + mh^2)\ddot{\theta} + (I_3 - I_2 - mh^2)\left(\frac{v_r \tan \delta}{b}\right)^2 \sin \theta \cos \theta - mgh \sin \theta = -mh \cos \theta \left( \frac{av_r}{b \cos^2 \delta} \dot{\delta} + \frac{v_r^2}{b} \tan \delta \right)$$



$$(I_1 + mh^2)\ddot{\theta} - mgh\theta = -\frac{mh}{b}(av_r \dot{\delta} + v_r^2 \delta)$$

---

# Non-linear vs. Linear Model



# State Space

---

- A derivative of the input is present in EOMs:

$$(I_1 + mh^2)\ddot{\theta} - mgh\theta = -\frac{mh}{b}(av_r\dot{\delta} + v_r^2\delta)$$

- A hand-derived *Controller Companion* form yields:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & \frac{mgh}{I_1 + mh^2} \\ 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} \delta$$

$$\theta = \begin{bmatrix} \frac{-mhav_r}{b(I_1 + mh^2)} & \frac{-mhv_r^2}{b(I_1 + mh^2)} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + [0]\delta$$

- But now the physical significance of the states is lost.
-

# Observer Companion Form

---

- Converting to this form yields:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{mgh}{I_1 + mh^2} & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} \frac{-mhav_r}{b(I_1 + mh^2)} \\ \frac{-mhv_r^2}{b(I_1 + mh^2)} \end{bmatrix} \delta$$

$$\theta = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \end{bmatrix} \delta$$

- Now, the output  $\theta$  **IS** the state  $x_1$
  - Specifying  $x_1(0)$  is specifying the initial bike roll angle,  $\theta(0)$ .
-

# Stability, Controllability, & Observability

---

- Unstable system:

$$s_{1,2} = \pm \sqrt{\frac{mgh}{I_1 + mh^2}}$$

- Controllability:

$$P = [B \quad AB] = \begin{bmatrix} -2.4 & -24.1 \\ -24.1 & -22.8 \end{bmatrix}$$

$$\text{rank}(P) = 2$$

- Observability:

$$N = [C^T \quad A^T C^T] = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

---



# Control Design Criteria

---

- From an initial roll angle of  $5^\circ$  and a constant velocity of 5m/s, meet the following performance criteria:

Overshoot  $< 1$  degree

Settling time  $< 2$  sec

Steer Input  $< 20$  degrees

---

# Full State Feedback & Regulator Design

---

- Letting the desired state vector be zero:

$$x_{error} = x_d - x = -x$$

- Now the control law reduces to

$$u(t) = -Kx(t)$$

- The close-loop state equations become

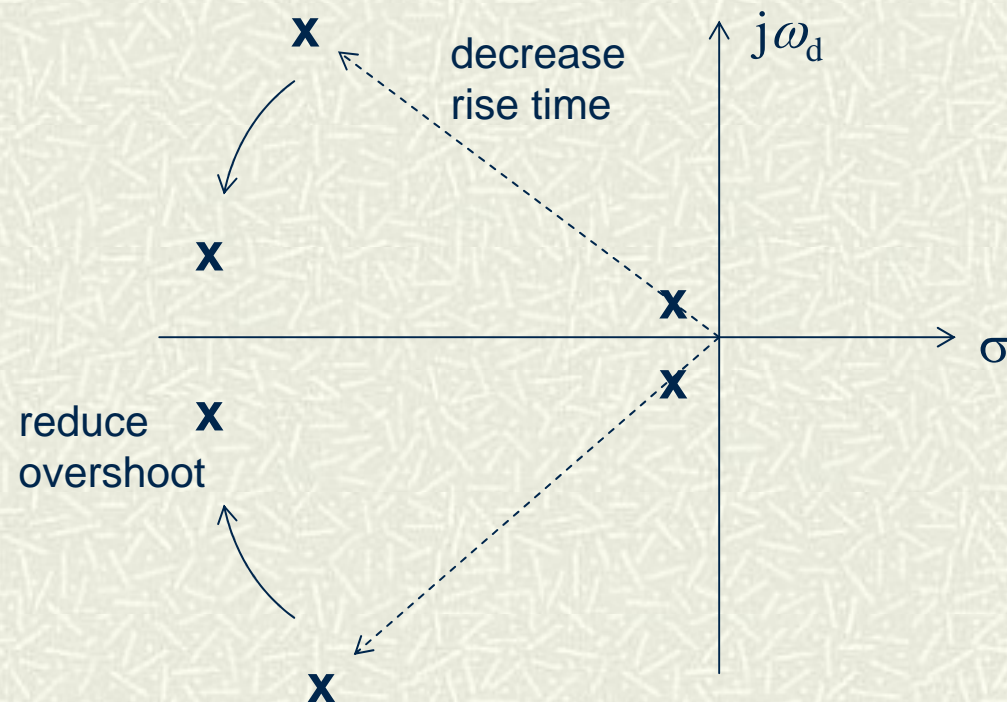
$$\dot{x}(t) = (A - BK)x(t)$$

- Consider design of the regulator by
    1. pole placement
    2. optimal control
-

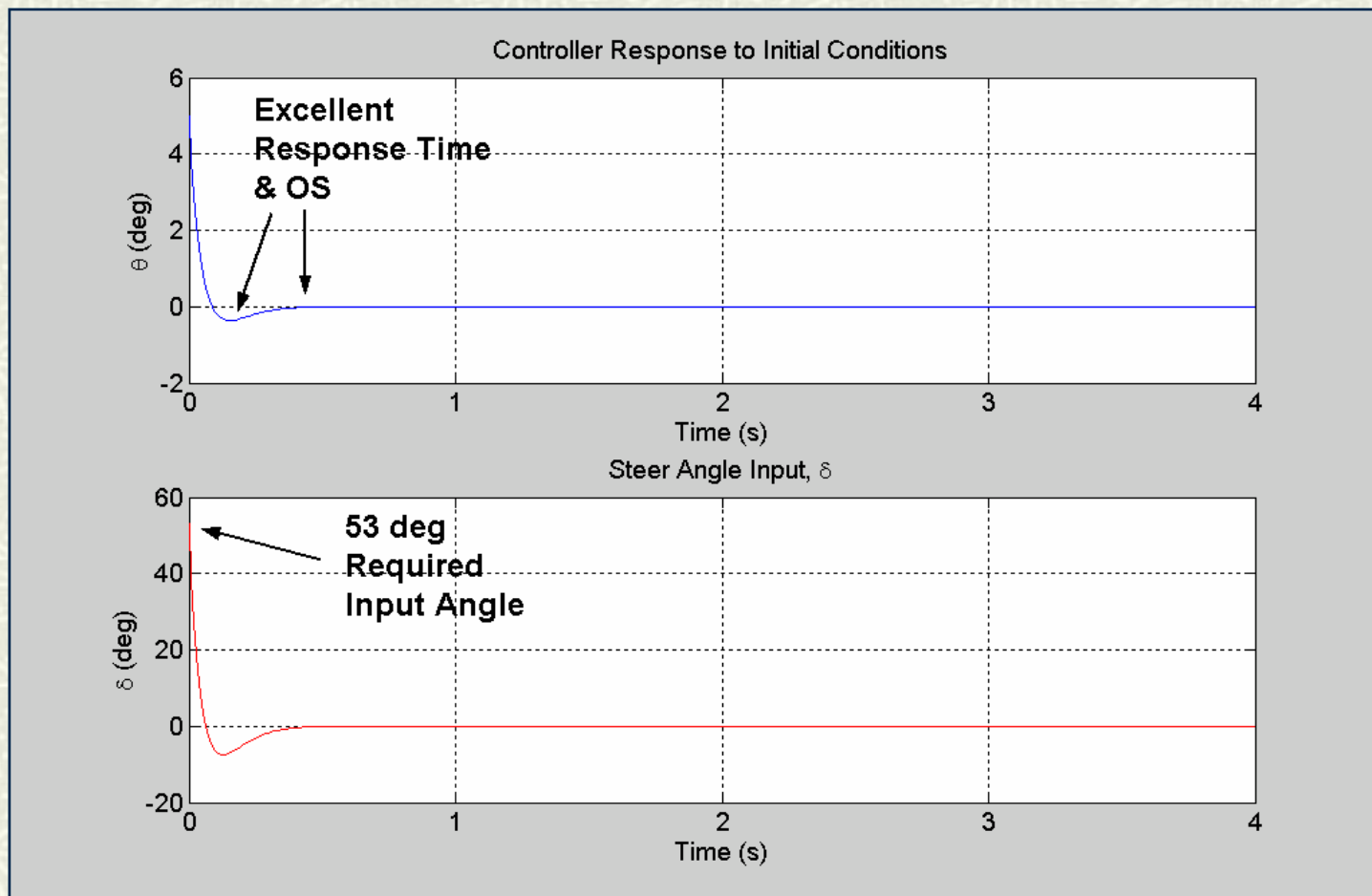
# Pole-Placement

- Performance requirements trade-offs:

Speed of Response vs. Overshoot



# Performance vs. Cost of Control



# Butterworth Pattern

---

- Place poles on a radius  $R$  centered @ origin.
- Poles obtained from the solution of:

$$(s / R)^{2n} = (-1)^{n+1}$$

where  $n$  is the order of the system

- For  $n=2$  the poles are the solutions of:

$$(s / R)^2 + (s / R)\sqrt{2} + 1 = 0$$

---

# Final Pole Placement Design

$$s_{1,2} = -3.01 \pm 3.01i$$



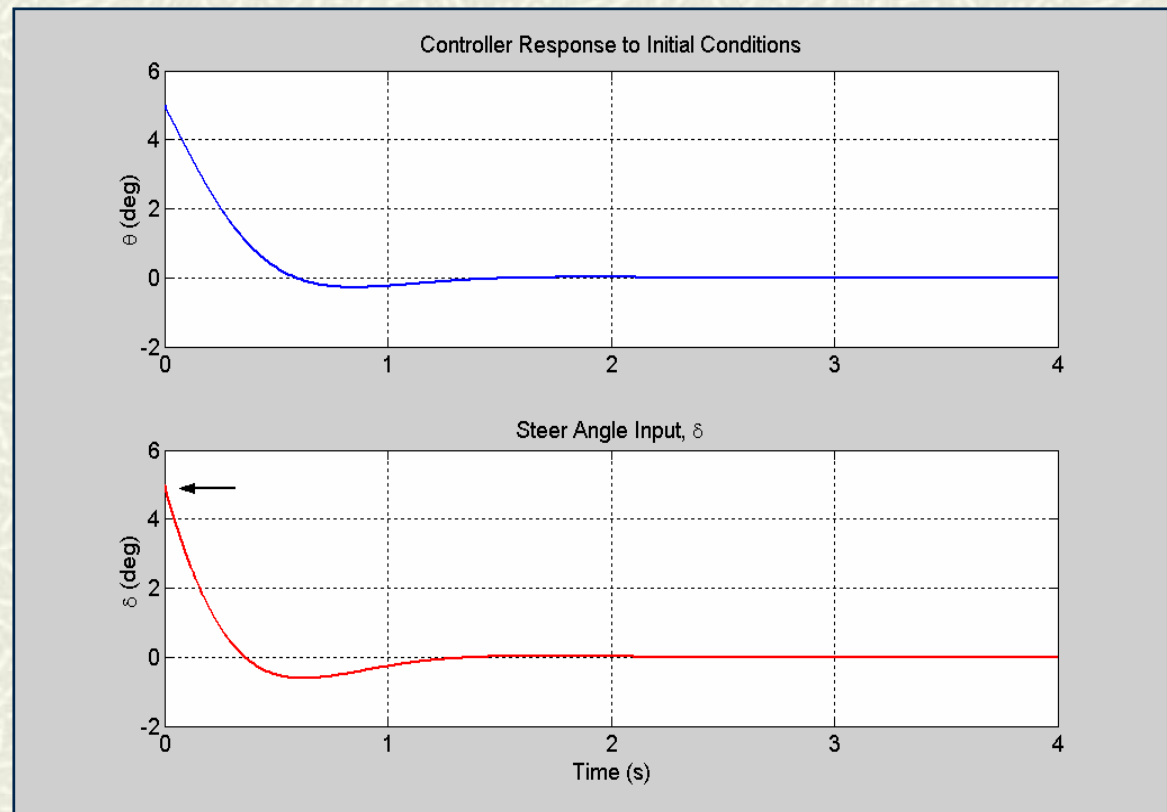
$$K = \begin{bmatrix} -1.0009 & -.1494 \end{bmatrix}$$

$$OS = .28 \text{ deg}$$

$$t_p = .8 \text{ sec}$$

$$T_s = 1.5 \text{ sec}$$

$$\delta_{\max} = 5 \text{ deg}$$



# LQR Control

---

- Physical meaning of states
- Physical limits on  $\delta$ :  $\pm 20^\circ$
- Physical limits on  $\theta$ :  $\pm 35^\circ$
- When  $x_1$  and  $\delta$  were weighted equally and  $x_2$  was weighted less the controller provided good results

$$J(u) = \int_0^T (\vec{x}^T R \vec{x} + \vec{u}^T \Lambda \vec{u}) dt$$

$$\vec{x} = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \quad \begin{aligned} x_1 &= \theta \\ x_2 &= \dot{\theta} + \frac{mhav_r}{b(I_1 + mh^2)} \delta \end{aligned}$$

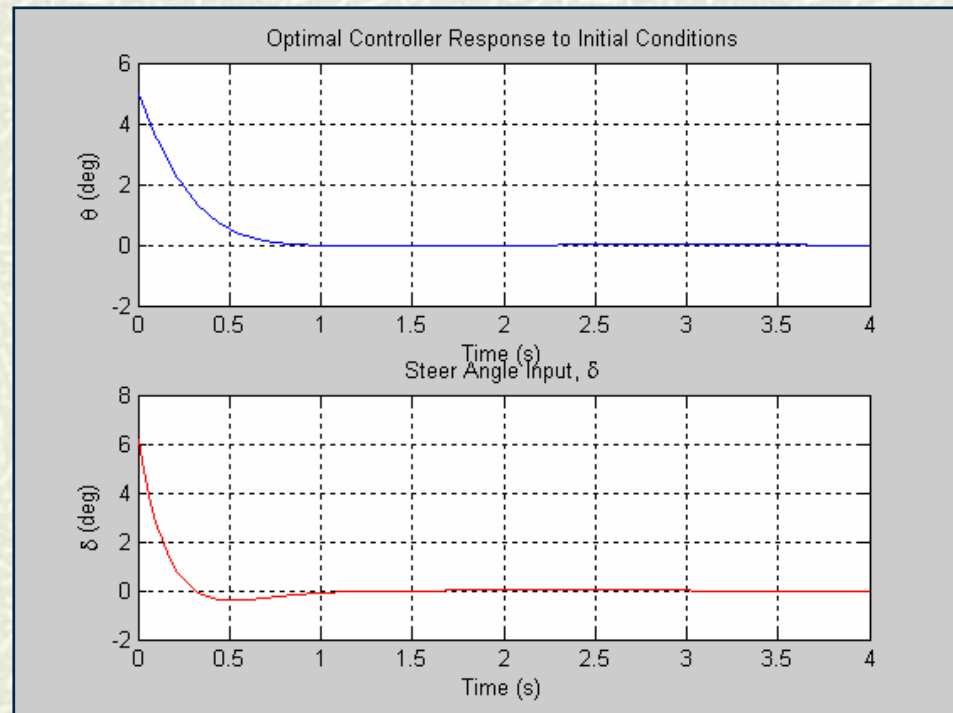
$$\vec{u} = [\delta]$$

---

# Underdamped Case

$$R = \begin{bmatrix} 1 & 0 \\ 0 & 0.01 \end{bmatrix} \quad \Lambda = 1 \quad \longrightarrow \quad K = \begin{bmatrix} -1.2308 & -.2539 \end{bmatrix}$$

- $s_{1,2} = -4.54 \pm 2.32i$
- OS = -0.02 deg
- $t_p = 1.1$  sec
- $T_s = 0.86$  sec
- $\delta_{\max} = 6.2$  deg

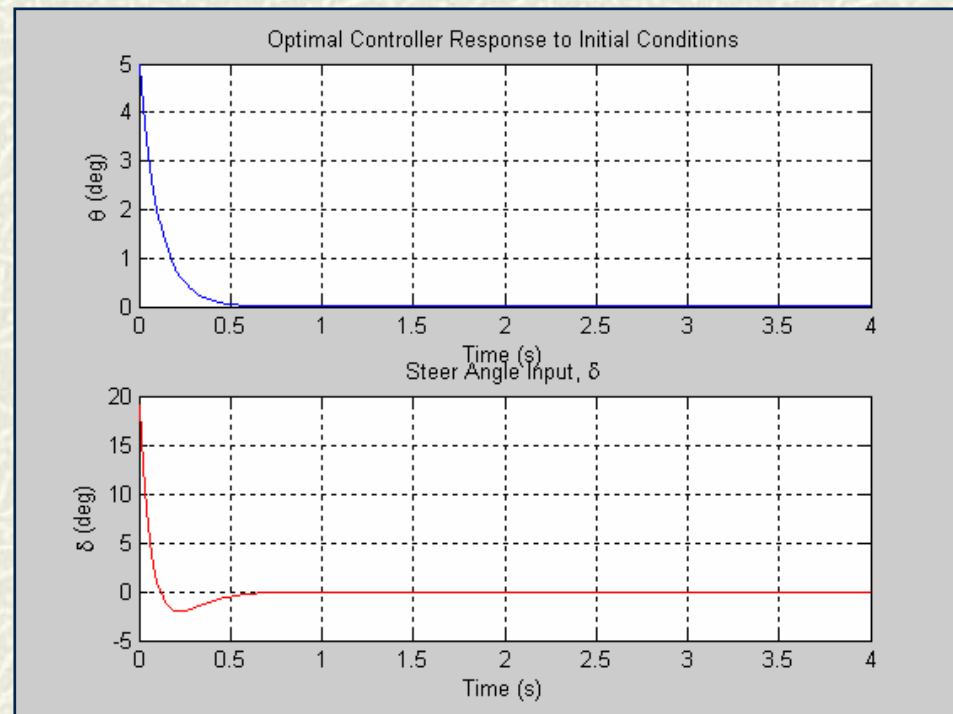




# Overdamped Case

$$R = \begin{bmatrix} 1 & 0 \\ 0 & 0.01 \end{bmatrix} \quad \Lambda = 0.07 \quad \longrightarrow \quad K = [-3.8165 \quad -.4153]$$

- $s_1 = -10.05$
- $s_2 = -9.14$
- $T_s = 0.60$  sec
- $\delta_{\max} = 19.1$  deg



# Classical Control Analysis

---

- Linear model (SISO system)

$$(I_1 + mh^2)\ddot{\theta} - mgh\theta = -\frac{mh}{b}(av_r\dot{\delta} + v_r^2\delta)$$

- Laplace transform of linear model equation.

$$(I_1 - mh^2)s^2\theta(s) = -\frac{mhv_r}{b}(as + v_r)\delta(s)$$

- Transfer function

$$\frac{\theta}{\delta} = \frac{-\frac{mhv_r}{b}(as + v_r)}{(I_1 + mh^2)s^2 - mgh}$$

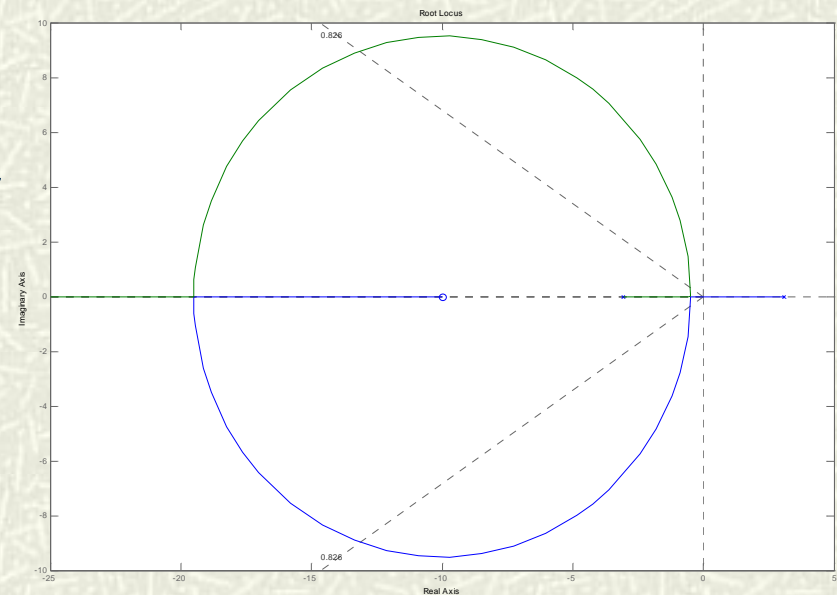
- As shown the system has a pole in RHP
  - The system is unstable in roll angle without some control of steering angle.
-

# Classical Control Analysis

For proportional control  
Routh- Hurwitz criteria to  
determine range of stability for  
K

$s^2$	1	9.42-K348
$s^1$	-87K	0
$s^0$	9.42-K348	0

$$\therefore K < 0$$



Root Locus of system proportionally controlled

# Classical Control Analysis

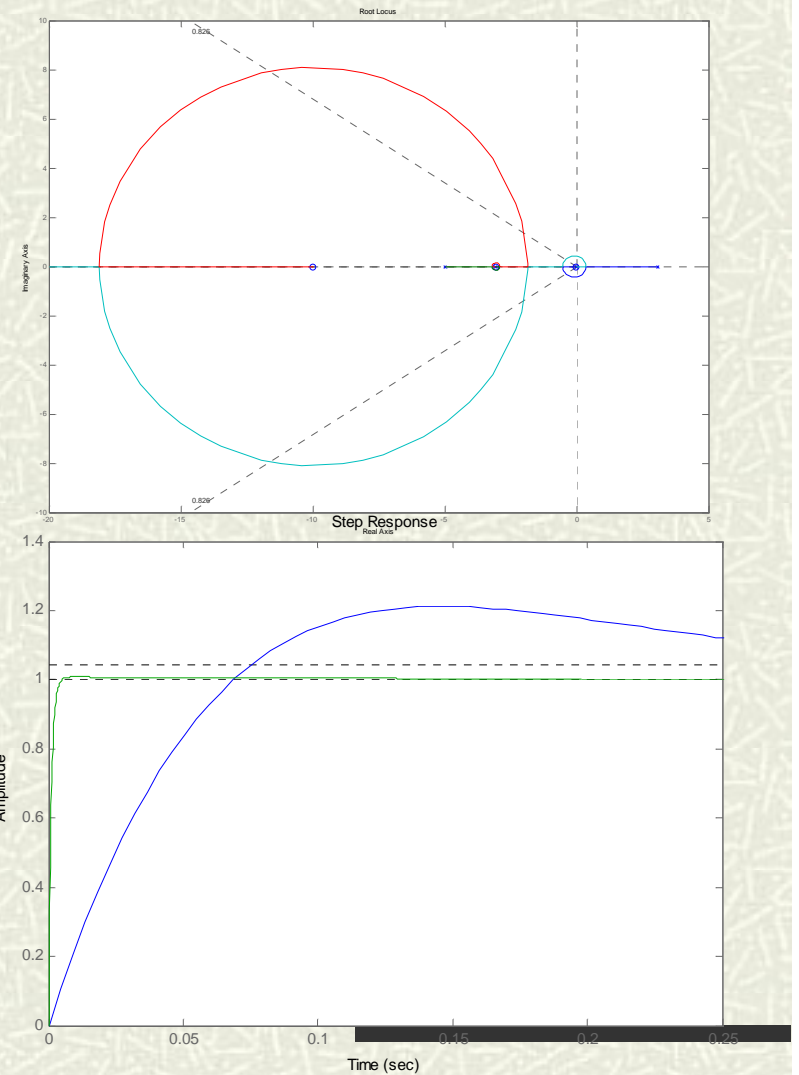
- Transient response improvement required: Lead controller design
- Decrease of steady state error : Lag controller design

## Controller TF

$$G_c = \frac{39.65((S + 3.07)(S + 0.05))}{(S + 5)(S + 0.01)}$$

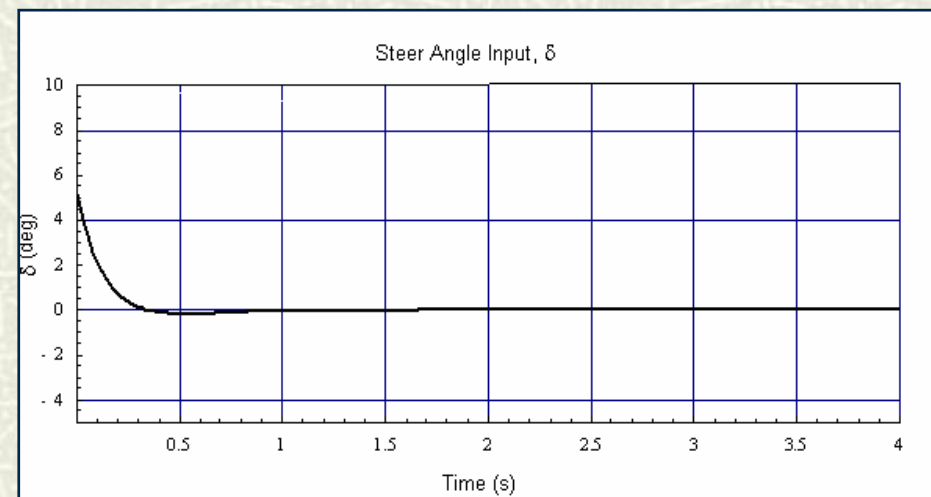
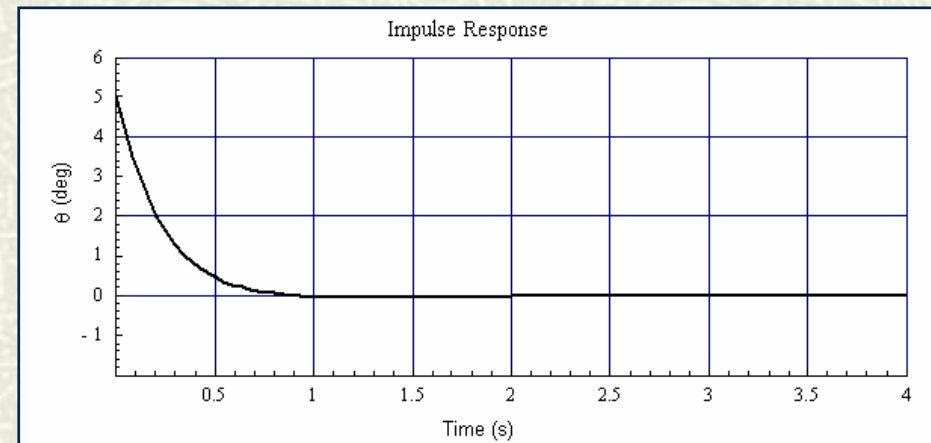
## Open loop transfer function

$$(G_c)(G) = \frac{39.65((S + 3.07)(S + 10)(S + 0.05))}{(S + 5)(S + 3.07)(S - 3.07)(s + 0.01)}$$



# Classical Control Analysis

- System response to a initial lean angle of 5 deg.
- $T_s=0.75$  sec
- $OS=-0.05$  deg
- $\delta_{max} = 5.1$  deg
- Dominant poles (CLTF)
- $S=-3.83+1.71j$
- $S=-3.83-1.71j$



# Conclusions

---

- Full-State Feedback is overkill for this SISO system
  - Classical methods provide more tools for design in SISO than using MIMO methods
  - LQR control methods are difficult to apply to when no physical limitations are in place
-

Questions???

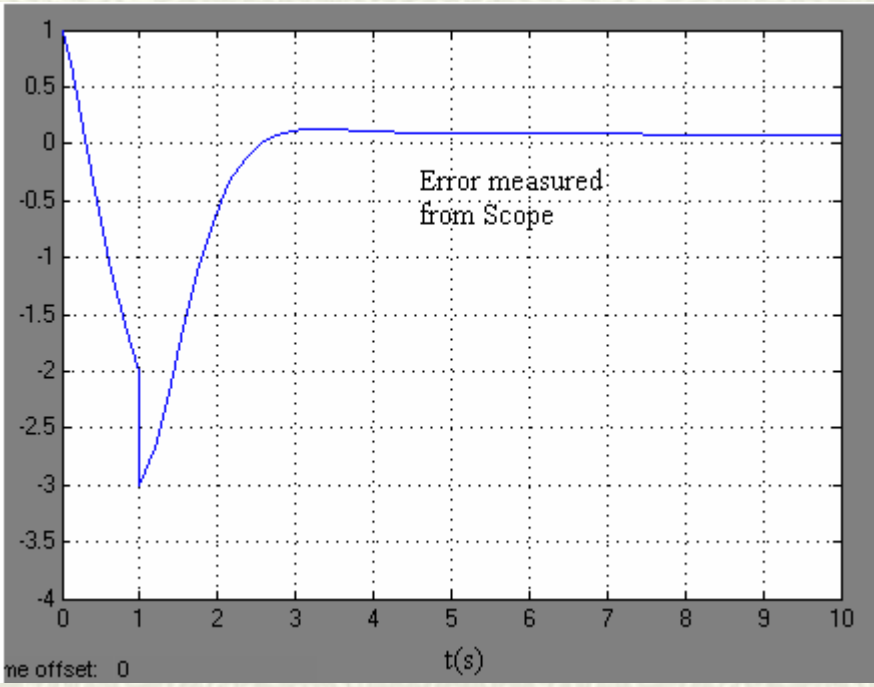


# Bicycle Parameters

---

- $I = 3.28 \text{ (kg}\cdot\text{m}^2)$
  - $m = 87 \text{ (kg)}$
  - $h = 1 \text{ (m)}$
  - $a = 0.5 \text{ (m)}$
  - $b = 1.0 \text{ (m)}$
  - $v_r = 5 \text{ (m/s)}$
-





This document was created with Win2PDF available at <http://www.daneprairie.com>.  
The unregistered version of Win2PDF is for evaluation or non-commercial use only.